



Issues in Experiments - part2

Facilities and wall shear stress measurements

Material prepared by Jean-Daniel Rüedi and Alessandro Talamelli





Topics



- High Reynolds number experiments
- Issues with facilities
- Wall shear stress measurements





HIGH REYNOLDS NUMBER EXPERIMENTS





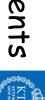


- High Reynolds number data required for fundamental turbulence research and flows of industrial interest
- Need for data with simultaneously
- High Reynolds number
- Resolution of all scales
- Well converged statistics





Minimum Reynolds number



Overlap region of the log law:

Approx: 300 < y+ < 0.15 R+

1 decade of y^+ in the Logarithmic region

- R+ > 20000

$$\frac{u}{u_\tau} = \frac{1}{\kappa} ln(y^+) + B$$





How to get high Re?





- High velocity (Limited by compressibility)
- Large size (Good resolution)
- High density (Pressurized facility)
- Low viscosity (Cryogenic facility)



High density experiments



- $D \approx 13$ cm
- L/D = 200
- High pressure (about 200 atm)
- Extremely large Re range : Re ≈ 5000 38 x 10⁶
- Very small viscous length scales



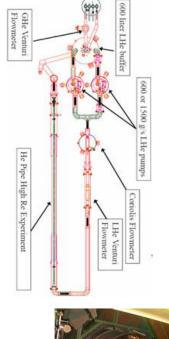
Zagarola and Smits, 1998



Cryogenic experiment



- axisymmetric jet and pipe geometries (HePipe, \mbox{R}^{\star} up to $3.10^{5}\mbox{)},$ very High Reynolds (up to $2*10^7$) in
- Very small viscous length scales









Large experiment



Long Pipe at CICLoPE, University of Bologna (in construction)

- Diameter: 0.9 m
- Length: 120 m
- Maximm R+: 50000
- Viscous length scale: > 12 µm
- Resolution of all scales
- Friction from pressure gradient
- Fully developed flow
- Mean velocity
- Higher moments





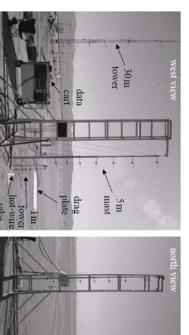
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Outdoor experiments



- SLTEST Facility, UTAH salt flats
- Very high Reynolds number
- Resolution of all scales
- Large variability due to the nature of the flow









ISSUES WITH FACILITIES





Flow quality



- Flow homogenity
- Flow angularity
- Turbulence intensity Noise level
- Temperature stability
- Velocity stability
- Pressure gradient control





Close loop Wind tunnel



- Test section can work at ambient pressure (easy access with sensors)
- High flow quality
- Optimal flow control
- Need of cooling to keep temperature constant





Open loop Wind Tunnel

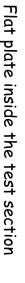


- Simpler and cheaper than close loop wind tunnel
- Limited control of the flow conditions
- Dust contamnination
- More power required for given speed
- Test section at ambient pressure (Blowing)
- Test section at low pressure (Suction)



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Flat plate versus wall flow



- Clean flow developing on the flat plate
- Easy conditioning of the surface
- Surface temperature generally equal to flow temperature
- Blockage under the flat plate
- Flow developing on the wall
- Easy access trough the wall
- Lot of space for bulky elements outside of the tunnel
- Use of the full height of the test section for the flow







Flow control



- Fan regulation
- Stable flow conditions over long time period for good statistics
- Velocity control
- Temperature control
- Reference velocity and temperature measurements





Inlet flow conditions



Turbulence manipulators

- Perforated plate
- Homogenization of the mean velocity
- Honeycomb
- Elimination of large eddies, flow angularity
- Screens and settling chamber
- Small scale mixing to homogenize the flow
- Decay of the small eddies





Contraction



- The design has a large impact on the flow quality
- Flat velocity profile
- No flow separation for low turbulence level
- No improvement of the flow quality for contraction ratio greater than 9:1
- Zero curvature at inlet & outlet

MATCHED-CUBIC -- FIFTH-ORDER

- Gentle curvature after the inlet
- Concave curvature destabilize the flow
- Danger of flow separation -> high rms

C=9

CENTER & Û ZERO LEVEL

2 cm ---

- Stronger curvature near the outlet
- Convex curvature stabilize the flow

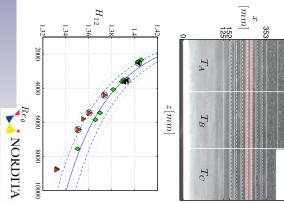


Boundary layer tripping.



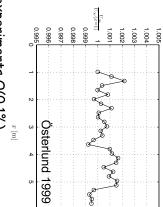
Orlu 2009

- origin of the turbulent BL Fix the transition point and the virtual
- Ensure span-wise flow homogeneity
- different velocities Ensure constant tripping position at
- of the tripping effect (Chauhan et al The shape factor H is a good indicator
- Issue for DNS simulation of boundary layer flows



Pressure gradient control

- Mobile roof panels (stream-wise and span-wise)
- Constant free-stream velocity
- Zero pressure gradient
- **Measurements**
- Pressure taps
- Velocity traverse



- Target pressure gradient for ZPG experiments O(0.1%)
- A test performed in the NDF at IIT (Nagib et al 2009) showed triction velocity of about 4%. that a pressure gradient of 4% results in a change of the





Blockage



- Asymmetric blockage due to traversing system, support, etc generate circulation around the flat plate
- The circulation increase the angle of attack and can lead to leading edge separation
- Control of the circulation using a flap at the trailing edge of the flat plate





Diffuser



- Long diffuser
- Prone to large scale separation
- Separation affect the flow quality in the test section
- Turbulence generators may have to be installed near the flap to avoid flow separation in the diffuser
- Detached flow from models can also lead to diffuser separation
- Split diffuser
- Create little more pressure drop that long diffuser
- Compact
- Less sensitive to inlet flow conditions
- Separation, if any, is localized, hence has less influence on the flow quality

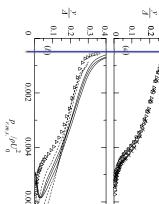




Noise level



- Noise source
- Fan noise
- Flow generated noise
- Noise reduction
- Sound absorbing material
- Acoustic mufflers
- Requirement for the MTL
- prms < 0.00015*q
- (84dB at 60 m/s)
- 1/10 of the minimum turbulent pressure fluctuation level



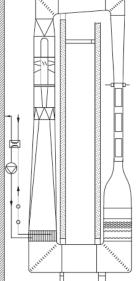
Tsuji et al. 2007



MTL at KTH, Stockholm 🚐



- Maximum velocity: 69 m/s (empty test section)
- Mean flow uniformity < 0.1%
- Turbulence intensity < 0.03%
- Temperature stability < ±0.1°C
- Honeycomb, 5 cleanable screens and 9:1 contraction
- Acoustic level 83 dB at 60 m/s
- Flat plate at mid-height
- Long diffuser



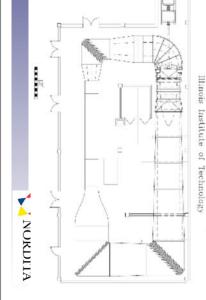




NDF at IIT, Chicago



- Test section $1 \times 1 \times 10$ m (H × W × L)
- Maximum velocity: 110 m/s
- Mean flow uniformity < 0.1%
- Turbulence intensity < 0.05%
- Motorized ceiling for pressure gradient control
- Honeycomb, screens and 9:1 contraction liagnostic Facility
- Flat plate at mid height
- 1:3 Split diffuser



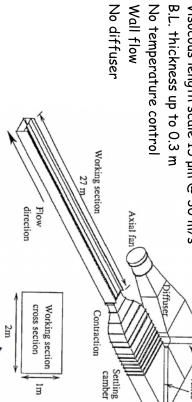


HRNBLWT in Melbourne



- Test section: $1 \times 2 \times 27 \text{ m} (H \times W \times L)$
- Maximum velocity: > 30 m/s
- Ret: 20'000 @ 30 m/s
- Visocous length scale 15 µm @ 30 m/s

- No diffuser





Long pipe at CICLoPE, Bologna



- **Expected characteristics**
- Test section: $0.9 \times 115 \text{ m} (D \times L)$
- Maximum velocity: > 65 m/s (R+ > 65'000, Power: 340 kW)
- Viscous length scale 11 µm @ 38 m/s (R+=40'000)
- Resolution of all scales with hot-wires
- Fully developed flow (Mean velocity, higher moments)
- Friction from pressure gradient
- Temperature stability $O(\pm 0.1^{\circ}C)$
- Velocity stability O(±0.1%)
- Acoustic level < 87dB (69 m/s)
- Honeycomb, 5 cleanable screens and 4:1 contraction







WALL SHEAR STRESS



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Direct techniques



- No hypothesis on the velocity profile
- Pressure drop in fully developed pipe (channel) flow
- Oil film interferometry
- Wall balances
- Momentum technique



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Indirect techniques



- Hypothesis on the velocity profile
- Calibration with a reference shear flow
- Preston tube
- Clauser plot, profile fit

 $\frac{u}{u_{\tau}} = \frac{1}{\kappa} ln(\frac{y \ u_{\tau}}{\nu}) + B$

- Wall fence
- Wall hot-wire, pulsed wire
- Wall film, MEMS film
- Micro pillar, micro fence
- Liquid crystals
- · ETC...



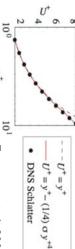
Near wall measurements



- Rely on the existence of a linear viscous sub-layer bellow y+ = 5 (3.5)
- Measurement techniques

 $U^{+} = y^{+} - \frac{1}{4}\sigma y^{+}$

- Hot-wires
- Optical techniques
- · (μ)LDA
- · (μ)PIV



Fransson et al. 2009

The viscous sub-layer is very thin at high Reynolds number, hence making measurements in this region fairly complicated



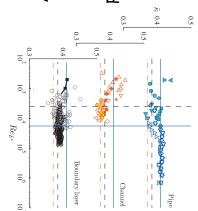
MAOIOE

Mean wall shear stress



- Crucial for the scaling in inner variables

 Von Karman "k" is NOT constant, hence profile based techniques should only be used
- Direct techniques are the only solution as long as one don't have a clear understanding of the behavior of the Karman "k"



with great care if at all

Nagib and Chauhan 2008

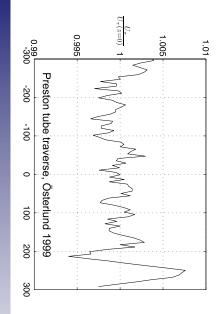




Wall shear stress variability



Span-wise wall shear stress variation A proper setup allows very good span-wise homogeneity Very sensitive to experiment setup







Momentum integral method



- Does not require fully developed flow
- Difficult to use in practice
- Limited accuracy

$$\frac{\overline{\tau}}{\int_{0}^{2} \frac{d\theta}{U_{\infty}^{2}}} = \frac{d\theta}{dx} + (H+2)\frac{\theta}{U_{\infty}}\frac{dU_{\infty}}{dx}$$



Wall balances



- Flush mounted elements
- Sensitive to alignment, gap size, pressure gradient and vibrations
- Sensing element size
- Large to be sensitive
- Small to measure local skin friction
- New development in MEMS balances
- Favorable scaling of errors at micro scale





Pressure drop



Balance between wall friction and pressure drop

$$\overline{\tau} = \frac{\Delta P}{2} \frac{R}{L}$$

- Only applicable in fully developed flows
- Correction for channel
- Care must be taken with the tap design for the static pressure measurement



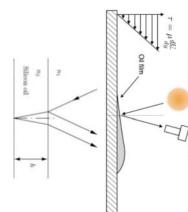


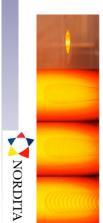
Oil film interferometry (OFI)



- Measurement of the thinning rate of an oil film
- Monochromatic light source
- Digital camera
- Glass surface
- Silicone oil
- Surface temperature sensor

$$\overline{\tau_w} \ k + \overline{\tau_w} \frac{h_0}{\Delta h} = \mu \ u_k \frac{2\sqrt{n^2 - \sin^2 \alpha}}{\lambda}$$







OFI - Measurement procedure



- temperature Independent calibration of the oil viscosity vs
- Spatial calibration with a target
- Acquisition of the images
- Analysis of the fringe spacing vs time



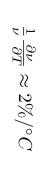
OFI - Oil viscosity calibration



- Thermo-regulated bath
- Capillary viscometer
- Optical barrier or stop watch
- Reference temperature sensor
- Accuracy ≈ 0.3%

225 230

±0.3%





OFI - Temperature measurements 👯



Evaluation of the surface temperature

$$\frac{\Delta T_{(oil-air)}}{\Delta T_{(oilfilm)}} \simeq O(10^3)$$

Thermistor	RTD (PT100)	Thermocouple	Туре
Resistor	Resistor	Junction voltage 1.0-2.2°C	Principle
0.1-0.3° <i>C</i>	0.03-0.3° <i>C</i>		Accuracy without calibration
Small	Large	Small	Size
Small Self heating	0.03-0.3°C Large (Self heating)		Size Danger



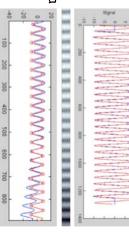


OFI Analysis method



- XT, Wavelength, Peak distance
- used correctly Very good agreement between the methods is obtained when
- formation of the oil film Initial transient due to the
- user-dependent and can lead to a fringes (XT method) is very The manual selection of the large scatter of the results







OFI - Time and spatial range



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- due to the formation of the oil film Transient apparent decrease of the wall shear stress
- Potential surface tension effect at the edge of the



Fluctuating wall shear stress

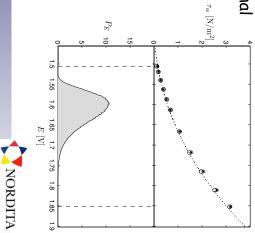


- Characteristics of the signal
- Long tail PDF
- $S \approx 1, F \approx 4$

$$au'/\overline{ au} pprox 0.4$$

Calibration range

$$0.3-3 \overline{\tau}$$

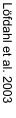


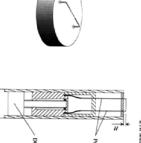
Wall wire devices

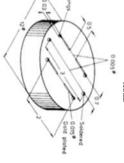


- Calibration in a reference flow
- Measurement of the mean and fluctuations
- Problems:
- Height of the sensor limit its use
- Heat transfer at the wal









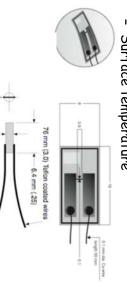


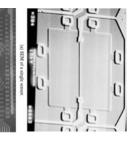


Wall film devices



- Calibration in a reference flow
- fluctuations Measurement of the mean and
- Problems:
- Heat transfer trough the surface
- Surface temperature









.15 mm (.006)

1.5 mm (.060) Dia

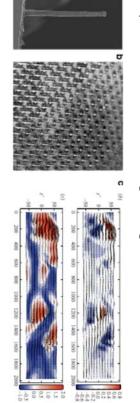




Micro pillar



- Institute of Aerodynamics, RWTH Aachen University
- pillars by the flow near the wal Principle: Optical measurement of the deformation of
- Static calibration in a reference flow
- Dynamic calibration using a magnetic field



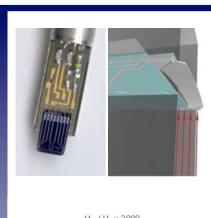


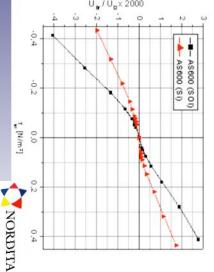


MEMS surface fence



Principle: Deformation of an element by the flow Static calibration in a reference flow Micro-sensor and actuator technology center TUB, Berlin



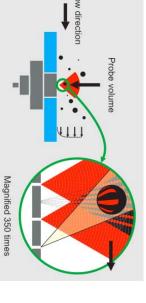


Micro-Optical sensor



Commercially available from MSE Frequency proportional to the velocity gradient Micro-Optical Sublayer Shear Stress Sensor, MSE

Measurement height: 75 μ m and 135 μ m









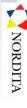
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