## Course Project



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## Jets and Wakes

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- Governing equations are just as for Blasius boundary layer
- Boundary Conditions:

Jet:  $u \to 0$  as  $y \to \pm \infty$ 

Wake: 
$$u 
ightarrow U_\infty$$
 as  $y 
ightarrow \pm \infty$ 

• For high Re, parallel flow can be assumed Jet:  $U = U(y) = sech^2(y)$ 

Wake:  $U = U(y) = 1 - \lambda sech^2(y)$ 



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- Disturbance equations are derived from Navier-Stokes equations by:
  - Decomposing velocities
  - Subtracting meanflow
  - Linearizing

$$\frac{\partial u_i}{\partial t} + \bar{U}_j \frac{\partial u_i}{\partial x_j} + u_j \frac{\partial \bar{U}_i}{\partial x_j} = -\frac{\partial p}{\partial x_i} + \frac{1}{Re} \nabla^2 u_i$$



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- Governing stability equations are the
  - Orr-Sommerfeld equationSquire equation
- They can be derived from the disturbance equations by assuming parallel flow

$$\left[\left(\frac{\partial}{\partial t} + \bar{U}\frac{\partial}{\partial x}\right)\nabla^2 - \bar{U}''\frac{\partial}{\partial x} - \frac{1}{Re}\nabla^4\right]v = 0$$

$$\left[\frac{\partial}{\partial t} + \bar{U}\frac{\partial}{\partial x} - \frac{1}{Re}\nabla^2\right]\eta = -\bar{U}'\frac{\partial v}{\partial y}$$

Boundary Conditions:  $v = \frac{\partial v}{\partial y} = \eta = 0$  as  $y \to \pm \infty$ 

• Base flow of the laminar Jet





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• Chebycheff collocation points not very suitable for this problem because resolution is worst in the middle



- High resolution of 201 points
- Height of box ( = 2\*Ymax) was varied

- Height of box was chosen as 15 which corresponds to Ymax=7.5
- Resolution was varied



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## • Neutral stability of laminar jet



• Real and imaginary parts of eigenfunctions for the critical values  $\alpha_c = 0.26$ , Re<sub>c</sub> = 4.3



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• Transient Growth ( $\alpha_c = 0.26$ ) stable, neutral and unstable case



• Optimal disturbance and optimal response for the stable case



• Optimal response and eigenmode for an unstable case (Re = 10)





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• Eigenmode and developed optimal disturbance are identical

• Pseudospectra of the stable case  $\alpha = 0.26$ , Re = 4.2



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• Sensitivity of eigenvalues for a stable and an unstable case



## • Optimal response to forcing for stable and unstable mode



- In the stable case the least stable frequency of the homogeneous problem is excited
- This is not the case for the unstable mode

- Conclusions
- Inflection point in profile, therefore the Jet is inviscidly unstable



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- Jet becomes unstable at very low Reynold numbers
- The operator matrix is almost normal and thus the eigenvectors are almost orthogonal
- Low Reynoldsnumbers quantitavely not relevant as flow is not parallel
- Laminar Jet would not be observable in reality

 Base flow of the laminar Wake for different λ







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• Neutral stability of laminar wake



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•  $\alpha_c = 0.4$ ,  $\text{Re}_c = 7.27$ 

• Eigenspectrum and least stable eigenmode for the neutral case  $\alpha_c = 0.4$ , Re<sub>c</sub> = 7.27



• Transient Growth, optimal disturbance and optimal response for the stable case  $\alpha = 0.4$ 



-0.6

0.1

0.2

0.3

|v|, |u|

0.4

0.5

0.6

0.7



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0.6

0.4

0.2

-0.2

-0.4

-0.6

0.2

0.3

. |v|, |u|

0.4

0.5

0.6

0.7

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• Eigenvalue sensitivity and Optimal response to time harmonic forcing for a stable case  $\alpha = 0.4$ 



• Least stable mode of homogeneous problem is excited most

- Conclusions
- The wake behaves similar to the jet



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- Jet becomes unstable at very low
   Reynold numbers → inviscidly unstable
- The operator matrix is almost normal and thus the eigenvectors are almost orthogonal
- Low Reynoldsnumbers quantitavely not relevant as flow is not parallel
- Wake becomes more stable for increasing  $\lambda$