

# A simple model for the turbulent shear stress.

Total average stress  $-P\delta_{ij} + \tau_{ij} = -\rho \overline{u_i u_j}$

turbulent stress  
Reynolds stress

Average Newtonian stress

$$\tau_{ij} = 2\mu E_{ij} \quad ; \quad E_{ij} = \frac{1}{2} \left( \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right)$$

Average turbulent stress model

$$-R_{ij} = -\overline{u_i u_j} = -\frac{2}{3} \bar{k} \delta_{ij} + 2\nu_T E_{ij}$$

$$\frac{2}{3} \bar{k} = \frac{\overline{u^2} + \overline{v^2} + \overline{w^2}}{3}$$

turbulent kinematic viscosity

Typically  $\nu_T \gg \nu = \mu/\rho$

Suppose  $\nu_T = v_T \lambda$

typical velocity fluctuation

$$v_T \sim \sqrt{\bar{k}}$$

Prandtl's mixing length on which fluid particle retains it's momentum

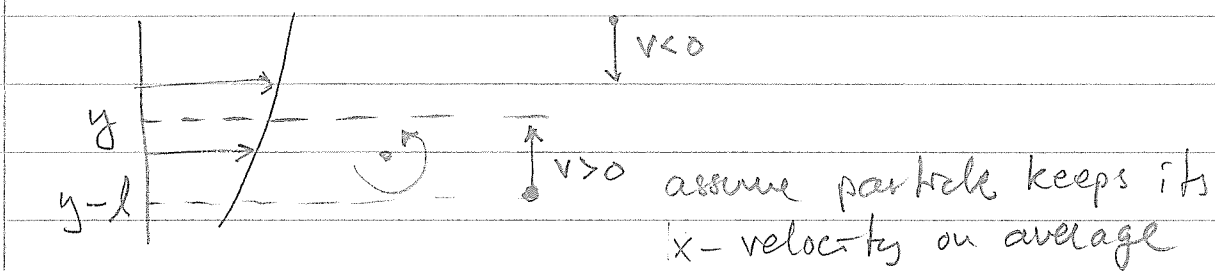
Compare kinetic gas theory  $\nu \approx \bar{c} \lambda$

$\bar{c}$  is molecular speed,  $\lambda$  is mean free path

$l(x, t)$  often taken from geometrical considerations.



# Typical velocity fluctuation $u_T$

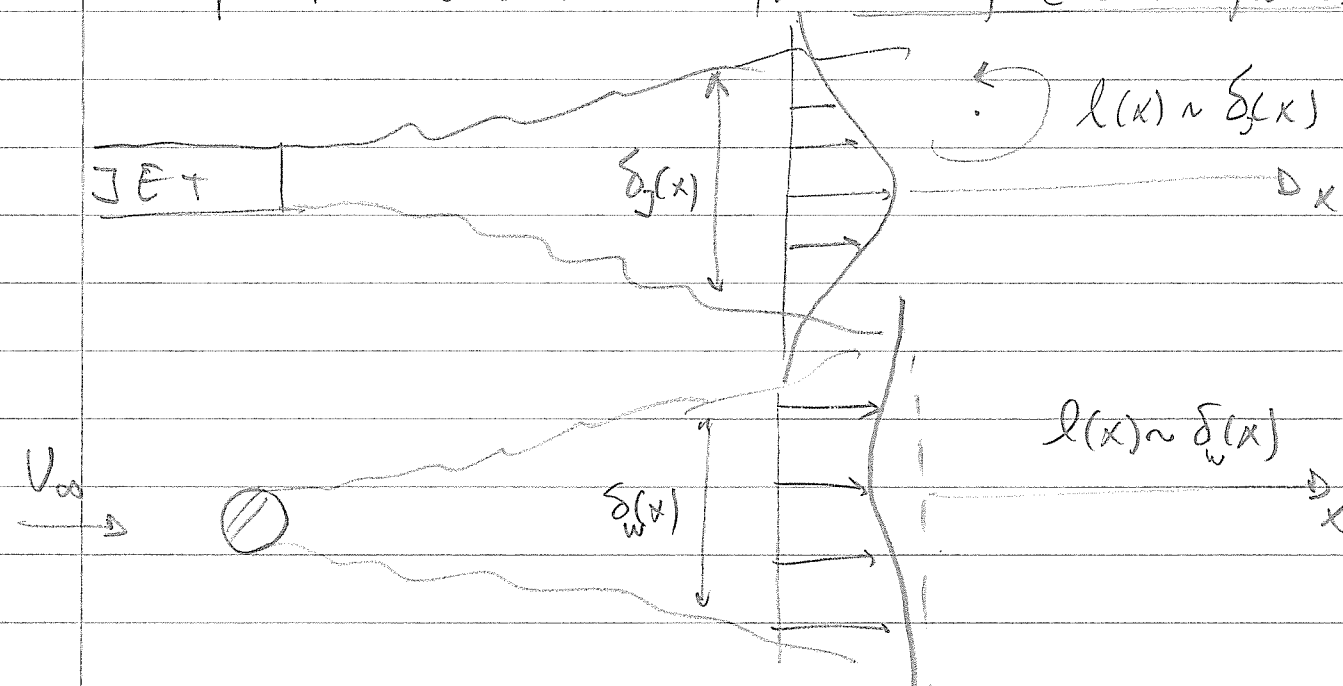


$$u_T \approx U(y-l) - U(y) \approx -\frac{dU}{dy} l$$

$$u_T \approx U(y+l) - U(y) \approx \frac{dU}{dy} l$$

$$-\overline{uv} \approx v_T l \frac{dU}{dy}$$

Simple model suitable for wall free shear flows.



One has found that the spreading rates

$$\frac{D}{Dt} \delta_j(x) \sim U_c(x) ; \quad \frac{D}{Dt} \delta_w(x) \sim \Delta U_c(x)$$

$$\sim U_c(x) \frac{d\delta_j}{dx} \quad \sim U_0 \frac{d\delta_w}{dx}$$

$$\Rightarrow \frac{d\delta_j}{dx} = \text{const.} \quad \frac{d\delta_w}{dx} = \frac{\Delta U_c(x)}{U_\infty} \text{ const.}$$

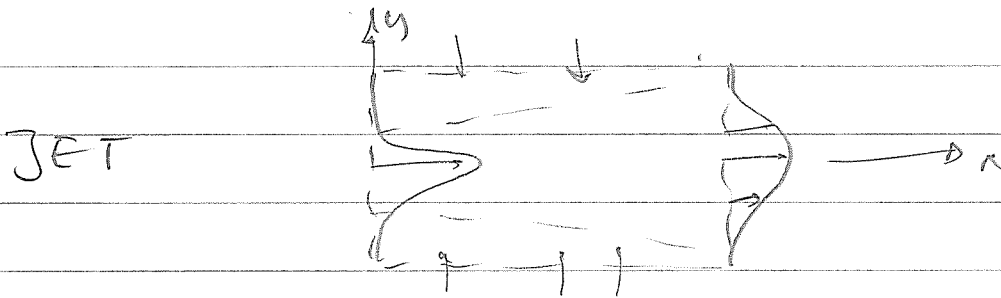
$$\approx \delta_j = \text{const.} \cdot x$$

Self-preservation far downstream

$$\text{JET} \quad U(x, y) = U_c(x) F_j(y/\delta_j(x))$$

$$\text{WAKE} \quad \Delta U(x, y) = \Delta U_c(x) F_w(y/\delta_w(x))$$

Combined with momentum theorem



$$\text{Momentum flux} \quad M = \int_{-\infty}^{\infty} \rho U^2 dy = \text{const.}$$

(although  $\int_{-\infty}^{\infty} \rho U dy \neq \text{const.}$ )

WAKE

$$U = U_\infty - \underbrace{\Delta U(x, y)}_{\ll U_\infty}$$

$$\text{Drag } D' = \rho U_\infty^2 \int_{-\infty}^{\infty} \frac{U}{U_\infty} \left(1 - \frac{U}{U_\infty}\right) dy = \rho U_\infty^2 \int_{-\infty}^{\infty} \left(1 - \frac{\Delta U}{U_\infty}\right) \frac{\Delta U}{U_\infty} dy$$

$$\approx \rho U_\infty^2 \int_{-\infty}^{\infty} \Delta U dy$$

So, JET

$$M = \int_{-a}^{\infty} \rho U_c^2 F_j(y) dy \delta_j(x) = \rho U_c^2(x) \delta(x) \underbrace{\int_{-a}^{\infty} F_j(y) dy}_{\text{const.}}$$

$$\Rightarrow \underbrace{\rho U_c^2(x) \delta(x)}_{\sim x} = M / \text{const.}$$

$$\Rightarrow U_c^2 = \frac{M}{\rho x} \quad U_c(x) = \sqrt{\frac{M}{\rho x}} \text{ const.}$$

L WAKE

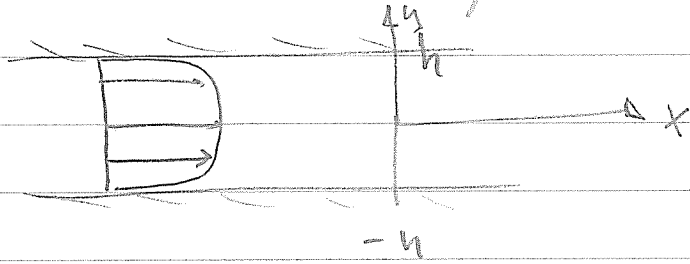
$$\begin{aligned} D' &= \rho U_\infty \int_{-a}^{\infty} \Delta U_c(x) F_w(y) dy \delta_w(x) = \\ &= \rho U_\infty \underbrace{\Delta U_c(x)}_{\sim U_\infty \frac{d\delta_w}{dx}} \delta_w(x) \underbrace{\int_{-a}^{\infty} F_w(y) dy}_{\text{const.}} \end{aligned}$$

$$\delta_w \frac{d\delta_w}{dx} = \frac{D'}{\rho U_\infty^2}$$

$$\Rightarrow \left. \begin{aligned} \delta_w(x) &= \sqrt{\frac{D' x}{\rho U_\infty^2}} \text{ const.} \end{aligned} \right\}$$

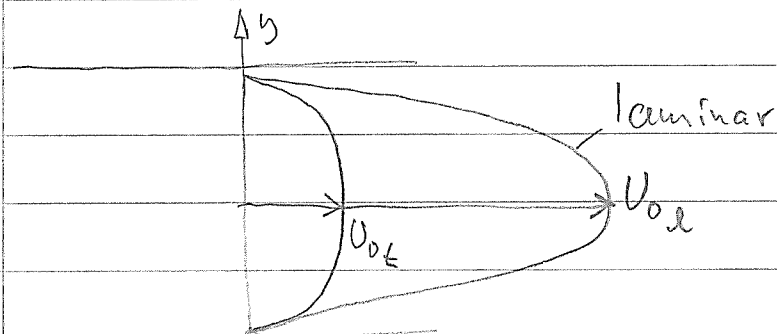
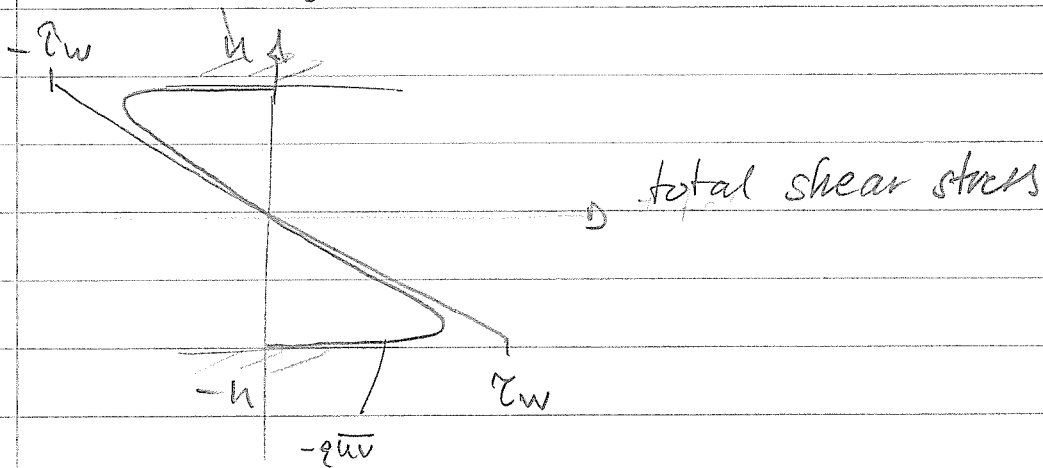
$$\left. \begin{aligned} \Delta U_c(x) &= U_\infty \sqrt{\frac{D'}{\rho U_\infty^2 x}} \text{ const.} \end{aligned} \right\}$$

## Turbulent channel flow



Fully developed  $\Rightarrow$  total shear stress =

$$= \mu \frac{dU}{dy} - \rho \overline{uv} = -\frac{y}{h} \tau_w$$

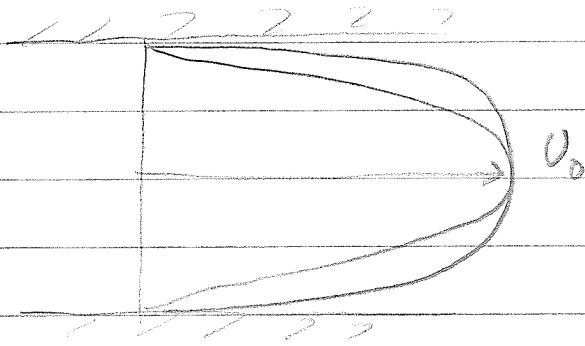


$$\overline{uv} = 0 \text{ at wall } \therefore \tau_w = \mu \frac{dU}{dy} (y = -h)$$

turbulent flow with same  $\tau_w$  as laminar case  
gives  $U_{0,t} < U_{0,l}$ .

turbulent flow with the same  $Re = \frac{U_0 h}{\nu}$

$\Rightarrow$  different  $\bar{\tau}_w$



$\bar{\tau}_w \text{ turb} \Rightarrow \bar{\tau}_w \text{ lam.}$

Structure of wall bounded shear flows

Introduce friction velocity  $u_* = \sqrt{\bar{\tau}_w / \rho}$

length  $l^+ = \nu / u_*$

Define  $y^+ = \frac{y+h}{l^+}$ ,  $U^+ = \frac{U}{u_*}$

if  $Re_+ = \frac{u_* h}{\nu} \gg 1$  then

